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| <b>SAMPLE</b>      | <b>ABNORMAL CHECKLIST</b> |               |
| <b>VERSION 1.0</b> |                           | <b>DA40-D</b> |

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| <b>DODAR PROCESS</b>   |
| Diagnosis / Options / Decide / Assign tasks and Action / Review decision |

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| <b>NITS MEMONIC</b>                               |
| Nature / Intentions / Time / Special Instructions |

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| <b>TAKE-OFF FROM A SHORT GRASS STRIP</b>   |
| BRAKES _____ APPLY<br>FLAPS _____ T/O<br>POWER LEVER _____ MAX<br>ELEVATOR _____ FULLY AFT<br>BRAKES _____ RELEASE<br>HOLD DIRECTION _____ USING RUDDER                                      |
| <i>In strong crosswinds steering can be augmented by use of the toe brakes. It should be noted, however, that this method increases the take-off roll, and should not generally be used.</i> |
| ELEVATOR _____ RELEASE SLOWLY<br><i>Allow airplane to lift off as soon as possible and increase speed at low level.</i>  |
| AIRSPEED _____ 66 KIAS (1150 kg)<br>FLAPS _____ UP<br>AIRSPEED _____ 75KIAS<br>LANDING LIGHT _____ AS REQD   |

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| <b>INDICATIONS OUTSIDE OF GREEN RANGE (OIL PRESS.)</b>   |
| HIGH OIL PRESSURE _____ CHECK OIL TEMP<br>COOLANT _____ CHECK COOLANT TEMP   |
| <i>If the temperatures are within the green range: - Expect wrong oil pressure indication. Keep monitoring temperatures. If the temperatures are not within the green range</i>  |
| POWER LEVER _____ REDUCE<br><i>Expect engine failure. Prepare for an emergency landing in accordance with 3.5.1 - EMERGENCY LANDING WITH ENGINE OFF.</i>   |
| LOW OIL PRESSURE _____ REDUCE POWER<br>POWER LEVER _____ REDUCE<br>MONITOR _____ CHECK OIL TEMP<br><i>Expect loss of oil with engine failure. Prepare for an emergency landing in accordance with 3.5.1 - EMERGENCY LANDING WITH ENGINE OFF.</i> |

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| <b>INDICATIONS OUTSIDE OF GREEN RANGE (RPM)</b>  |
| HIGH RPM _____ REDUCE POWER<br>RPM _____ GREEN RANGE   |
| <i>An RPM in the yellow range is permissible for a short time if required, e.g. for go-around. If the available power is too low to continue a safe flight, perform a precautionary landing on the nearest airfield in accordance with 4B.1 - PRECAUTIONARY LANDING.</i> |

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| <b>INDICATIONS OUTSIDE OF GREEN RANGE (GEARBOX)</b>                            |
| GEARBOX TEMP _____ HIGH<br>POWER LEVER _____ REDUCE<br>AIRSPEED _____ INCREASE |

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| <b>INDICATIONS OUTSIDE OF GREEN RANGE (COOLANT)</b>   |
| HIGH COOLANT TEMP _____ CHECK COOLANT (WATER LEVEL)<br>CAUTION LIGHT (WATER LEVEL) _____ OFF<br>DURING CLIMB _____ REDUCE POWER<br>INCREASE AIRSPEED _____ +10KIAS<br>DURING CRUISE _____ REDUCE POWER<br>INCREASE AIRSPEED _____ +10KIAS<br><i>If the coolant temperature does not return to the green range, perform a precautionary landing on the nearest airfield in accordance with 4B.1 - PRECAUTIONARY LANDING.</i> |
| CAUTION LIGHT (WATER LEVEL) _____ ON<br>POWER LEVER _____ REDUCE<br><i>Expect loss of coolant. A further increase in coolant temperature must be expected. Prepare for an emergency landing in accordance with 3.5.1 - EMERGENCY LANDING WITH ENGINE OFF</i>  |
| LOW COOLANT TEMP _____ CHECK COOLANT (WATER LEVEL)<br><i>During an extended descent from high altitudes with a low power setting coolant temperature may decrease.</i><br>POWER LEVER _____ REDUCE<br><i>Expect loss of coolant. A further decrease in coolant temperature must be expected. Prepare for an emergency landing in accordance with 3.5.1 - EMERGENCY LANDING WITH ENGINE OFF.</i>                             |

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| <b>FUEL TEMPERATURE</b>  |
| FUEL TEMP _____ HIGH<br>POWER LEVER _____ REDUCE<br>AIRSPEED _____ INCREASE<br><i>Increased fuel temperature can occur when the fuel quantity in the main tank is low. The fuel temperature can be decreased by transferring fuel from the auxiliary to the main tank.</i> |
| FUEL TEMP _____ LOW<br>POWER LEVER _____ INCREASE<br>AIRSPEED _____ REDUCE<br><i>If the fuel cooler is in operation (baffle removed): - Select lower flight altitude, if possible.</i>   |

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| <b>INDICATIONS OUTSIDE OF GREEN RANGE (OIL TEMP)</b>  |
| HIGH OIL TEMP _____ CHECK OIL PRESSURE<br>OIL PRESSURE _____ LOW<br>POWER LEVER _____ REDUCE<br><i>Expect loss of oil with engine failure. Prepare for an emergency landing in accordance with 3.5.1 - EMERGENCY LANDING WITH ENGINE OFF.</i> |
| <i>If the oil pressure is within the green range</i>  |
| POWER LEVER _____ REDUCE<br>AIRSPEED _____ INCREASE<br>MONITOR _____ OIL TEMP   |
| <i>If the oil temperature is low</i>  |
| POWER LEVER _____ INCREASE<br>AIRSPEED _____ REDUCE<br>MONITOR _____ OIL TEMP   |

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| <b>FAILURE IN FLAP OPERATING SYSTEM</b>   |
| FAILURE _____ CHECK FLAP POSITION<br>AIRSPEED _____ WHITE ARC<br>RECHECK _____ SWITCH POSITION                                      |
| ONLY UP AVAIL _____ 73 KIAS<br><i>Land at a flat approach angle, use power lever to control airplane speed and rate of descent.</i> |
| ONLY TO AVAIL _____ 73 KIAS<br><i>Land at a flat approach angle, use power lever to control airplane speed and rate of descent.</i> |
| ONLY LDG AVAIL _____ PERFORM NORMAL LANDING   |

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| <b>POST-FLIGHT REVIEW AIDE-MEMOIRE</b>   |
| What happened and why / Was the outcome positive or not / How do we repeat or avoid / Impact on Safety / Were SOP's followed / What are the learning points / Further action require |

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| <b>CERTAIN AIRSPEEDS (IN EMERGENCIES)</b>  |  |
| Engine failure after take-off (FLAPS T/O)  | 850kg – 59KT 1000kg – 66KT 1150kg – 72KT |
| Airspeed for best glide angle (FLAPS UP)   | 850kg – 60KT 1000kg – 68KT 1150kg – 73KT |
| Emergency Landing with ENG OFF (FLAPS UP)  | 850kg – 60KT 1000kg – 68KT 1150kg – 73KT |
| Emergency Landing with ENG OFF (FLAPS T/O) | 850kg – 59KT 1000kg – 66KT 1150kg – 72KT |
| Emergency Landing with ENG OFF (FLAPS LDG) | 850kg – 58KT 1000kg – 63KT 1150kg – 71KT |

|                    |                           |               |
|--------------------|---------------------------|---------------|
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**DODAR PROCESS**  
Diagnosis / Options / Decide / Assign tasks and Action / Review decision

**NITS MEMONIC**  
Nature / Intentions / Time / Special Instructions

**ELECTRICAL SYSTEM – LOW VOLTS**

LOW VOLTAGE \_\_\_\_\_ ON GROUND  
 CIRCUIT BREAKERS \_\_\_\_\_ CHECK  
 POWER LEVER \_\_\_\_\_ INCREASE  
 CAUTION LIGHT ON \_\_\_\_\_ TERMINATE FLIGHT

*This caution is indicated when the normal on-board voltage (14 V) drops below 12.6 V.  
 Possible reasons are: - A fault in the power supply. - RPM too low.*

LOW VOLTAGE \_\_\_\_\_ IN AIR  
 CIRCUIT BREAKERS \_\_\_\_\_ CHECK  
 ELECTRICAL EQUIPMENT \_\_\_\_\_ NON ESSENTIAL OFF  
 CAUTION LIGHT ON \_\_\_\_\_ CONSIDER ALTERNATOR FAILURE

*'Low Voltage' Caution During Landing - Follow LOW VOLTAGE (ON GROUND) after landing.*

**LOW FUEL**

FUEL TRANSFER PUMP \_\_\_\_\_ ON  
 FUEL QUANTITY \_\_\_\_\_ CHECK

*As soon as the amount of usable fuel in the main tank is less than 3 US gal (+2/-1 US gal), a caution message is displayed.*

*If the caution light does not extinguish: Expect loss of fuel. - Be prepared for an emergency landing. - Proceed in accordance with 3.5.1 - EMERGENCY LANDING WITH ENGINE OFF.*

**ELECTRICAL SYSTEM – ECU A**

ON GROUND \_\_\_\_\_ TERMINATE FLIGHT  
 ECU A DURING FLIGHT \_\_\_\_\_ PRESS ECU TEST  
*Hold for more than 2 seconds to reset the caution message. If reset, continue the flight.*  
 ECU A CAUTION \_\_\_\_\_ LAND ASAP

*The engine must be serviced after landing.*

**FAILURES ON AUX ENGINE DISPLAY**

HIGH ELECTRICAL LOAD \_\_\_\_\_ REDUCE CONSUMPTION  
 ELECTRICAL EQUIPMENT \_\_\_\_\_ NON ESSENTIAL OFF

*If the problem does not clear itself*

LAND \_\_\_\_\_ ASAP

LOW VOLTAGE \_\_\_\_\_ CHECK CIRCUIT BREAKERS  
 ELECTRICAL EQUIPMENT \_\_\_\_\_ NON ESSENTIAL OFF

*If 'Low voltage' is still indicated on the AED 125 - Follow procedure in 4B.3.4 - ALTERNATOR FAILURE (ALTERNATOR)*

HIGH VOLTAGE \_\_\_\_\_ LAND ASAP

**ELECTRICAL SYSTEM – ECU B**

ON GROUND \_\_\_\_\_ TERMINATE FLIGHT  
 ECU B DURING FLIGHT \_\_\_\_\_ PRESS ECU TEST  
*Hold for more than 2 seconds to reset the caution message. If reset, continue the flight.*  
 ECU B CAUTION \_\_\_\_\_ LAND ASAP

*The engine must be serviced after landing.*

**PRECAUTIONARY LANDING**

SYSTEM FAILURE/WEATHER/MEDICAL OR OTHER \_\_\_\_\_ DODAR PROCESS  
 OCCURRENCE IN FLIGHT \_\_\_\_\_ NITS MEMONIC  
 RADIO \_\_\_\_\_ ADVICE ATC  
 NORMAL OPS \_\_\_\_\_ REFER CHECKLIST

*A landing of this type is only necessary when there is a reasonable suspicion that due to fuel shortage, weather conditions, or at nightfall the possibility of endangering the airplane and its occupants by continuing the flight cannot be excluded.*

**ELECTRICAL SYSTEM – ALTERNATOR**

ALTERNATOR CAUTION \_\_\_\_\_ ILLUMINATED/BLINKING  
 CIRCUIT BREAKERS \_\_\_\_\_ CHECK  
 ESSENTIAL BUS \_\_\_\_\_ ON  
 ELECTRICAL EQUIPMENT \_\_\_\_\_ NON ESSENTIAL OFF  
 LAND \_\_\_\_\_ ASAP

**POST-FLIGHT REVIEW AIDE-MEMOIRE**

What happened and why / Was the outcome positive or not / How do we repeat or avoid / Impact on Safety / Were SOP's followed / What are the learning points / Further action require

**ELECTRICAL SYSTEM – ENGINE**

ENGINE FAILURE \_\_\_\_\_ CHECK  
 CED 125 \_\_\_\_\_ CHECK  
 AED 125 \_\_\_\_\_ CHECK

*If an indication either on the CED 125 or AED 125 is near the end of the green range, it may happen that it switches over to the yellow or red range for a short time. This will also cause the ENGINE caution light to illuminate*

*If an indication either on the CED 125 or AED 125 is outside of the green range, proceed in accordance with 4B.2 - INSTRUMENT INDICATIONS OUTSIDE OF THE GREEN RANGE.*

**CERTAIN AIRSPEEDS (IN EMERGENCIES)**

|  |  |
|--|--|
| Engine failure after take-off (FLAPS T/O)  | 850kg – 59KT 1000kg – 66KT 1150kg – 72KT |
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**ELECTRICAL SYSTEM – PITOT**

PITOT \_\_\_\_\_ CHECK ON

*If In icing conditions EXPECT LOSS OF static INSTRUMENTS.*

ALTERNATE STATIC \_\_\_\_\_ OPEN

*The Pitot heating caution message is displayed when the Pitot heating is switched off, or when there is a failure of the Pitot heating system. Prolonged operation of the Pitot heating on the ground can also cause the Pitot heating caution message to be displayed. In this case it indicates the activation of the thermal switch, which prevents overheating of the Pitot heating system on the ground. This is a normal function of the system. After a cooling period, the heating system will be switched on again automatically.*